

Towards a "concurrent design facility (CDF)" for nanosatellites: a back-of-the-envelope calculation of the overall energy budget of OUFTI-1

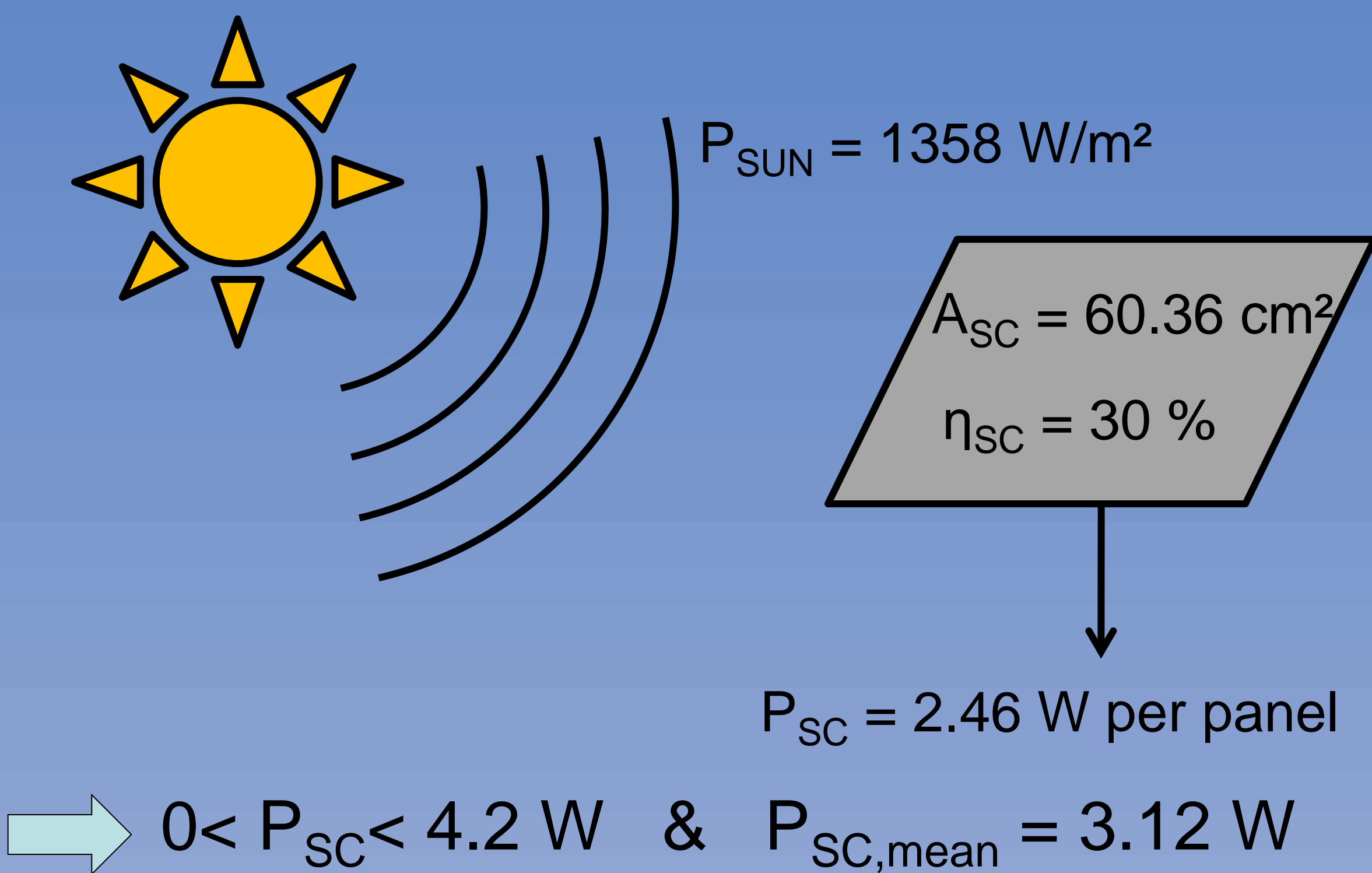


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Introduction

We give a very compact account of the overall energy budget of the OUFTI-1 nano satellite, thereby providing a useful guide for future designs, as well as an explanation of why a nanosatellite can, energetically, survive and do useful tasks in space.

Energy provider



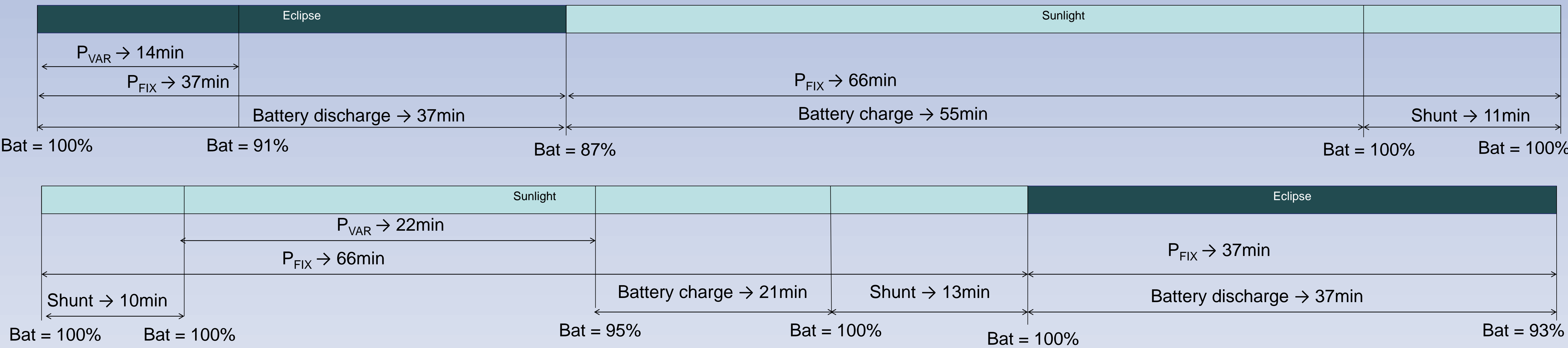
Energy budget

Check that EPS is correctly designed to support D-STAR communications

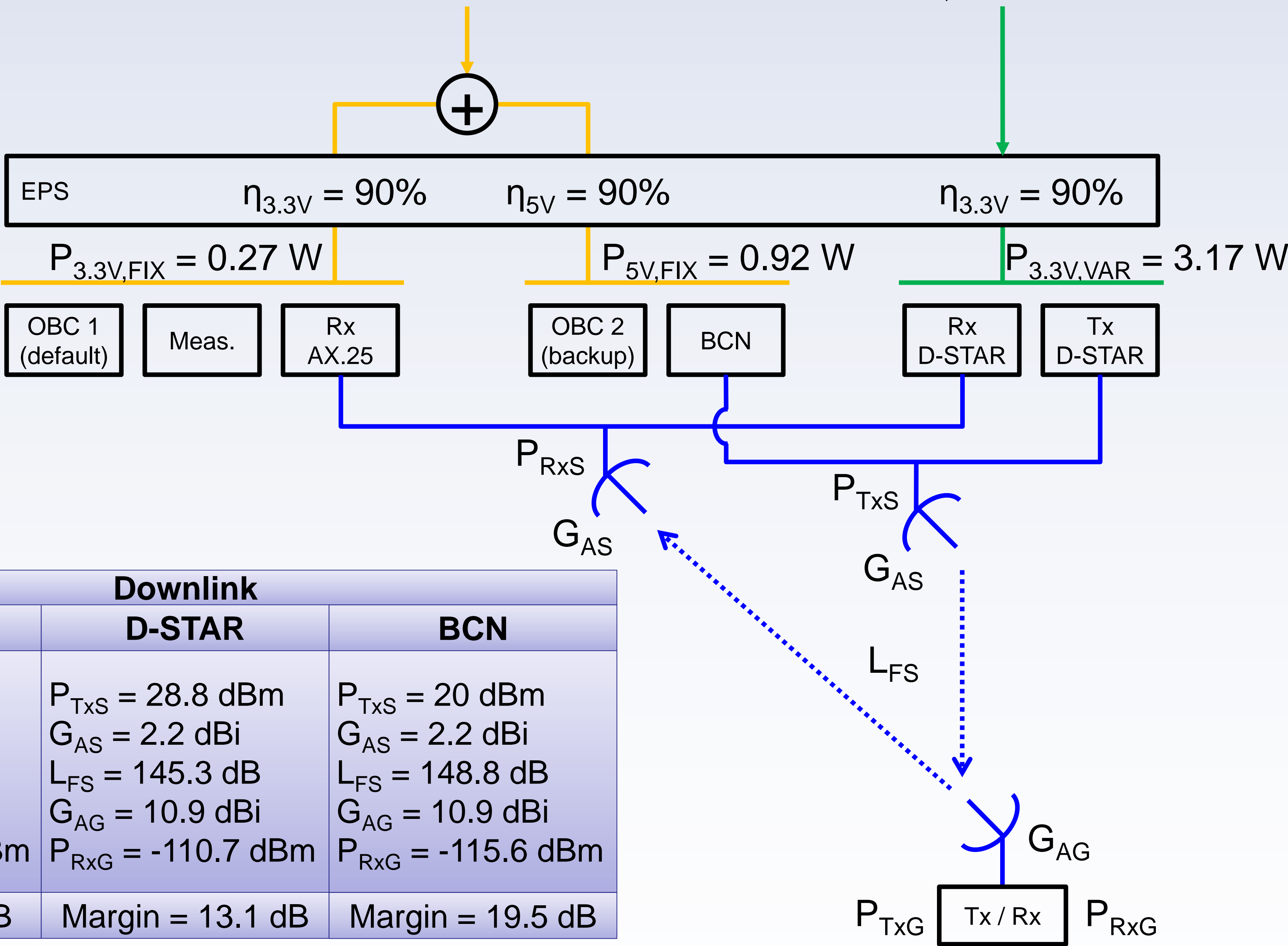
$$E_{ORB,prov} = E_{B,init} + P_{SC} T_{SUN}$$
$$E_{ORB,cons} = E_{B,final} + P_{FIX} T_{ORB} + P_{VAR} T_{COM}$$
$$T_{ORB} = T_{SUN} + T_{ECL}$$

Worst-case orbit:

$$T_{ORB} = 103 \text{ min}$$
$$T_{SUN} = 66 \text{ min}$$
$$T_{ECL} = 37 \text{ min}$$



$$P_{FIX} = P_{3.3V,FIX}/\eta_{3.3V} + P_{5V,FIX}/\eta_{5V} = 1.32 \text{ W} \quad P_{VAR} = P_{3.3V,VAR}/\eta_{3.3V} = 3.52 \text{ W}$$



EPS = Electrical Power Supply
OBC = On-Board Computer
BCN = Beacon
Meas. = Measurements
Rx = Reception
Tx = Transmission

Uplink		Downlink		
AX.25	D-STAR	AX.25	D-STAR	BCN
$P_{TxG} = 50 \text{ dBm}$	$P_{TxG} = 44 \text{ dBm}$	$P_{TxS} = 28.1 \text{ dBm}$	$P_{TxS} = 28.8 \text{ dBm}$	$P_{TxS} = 20 \text{ dBm}$
$G_{AG} = 12.8 \text{ dBi}$	$G_{AG} = 12.5 \text{ dBi}$	$G_{AS} = 2.2 \text{ dBi}$	$G_{AS} = 2.2 \text{ dBi}$	$G_{AS} = 2.2 \text{ dBi}$
$L_{FS} = 153.4 \text{ dB}$	$L_{FS} = 154.8 \text{ dB}$	$L_{FS} = 143.9 \text{ dB}$	$L_{FS} = 145.3 \text{ dB}$	$L_{FS} = 148.8 \text{ dB}$
$G_{AS} = 2.2 \text{ dBi}$	$G_{AS} = 2.2 \text{ dBi}$	$G_{AG} = 18.5 \text{ dBi}$	$G_{AG} = 10.9 \text{ dBi}$	$G_{AG} = 10.9 \text{ dBi}$
$P_{RxS} = -100 \text{ dBm}$	$P_{RxS} = -104.4 \text{ dBm}$	$P_{RxG} = -105.4 \text{ dBm}$	$P_{RxG} = -110.7 \text{ dBm}$	$P_{RxG} = -115.6 \text{ dBm}$
Margin = 16.9 dB	Margin = 21.9 dB	Margin = 7.5 dB	Margin = 13.1 dB	Margin = 19.5 dB

G = Ground, S = Satellite